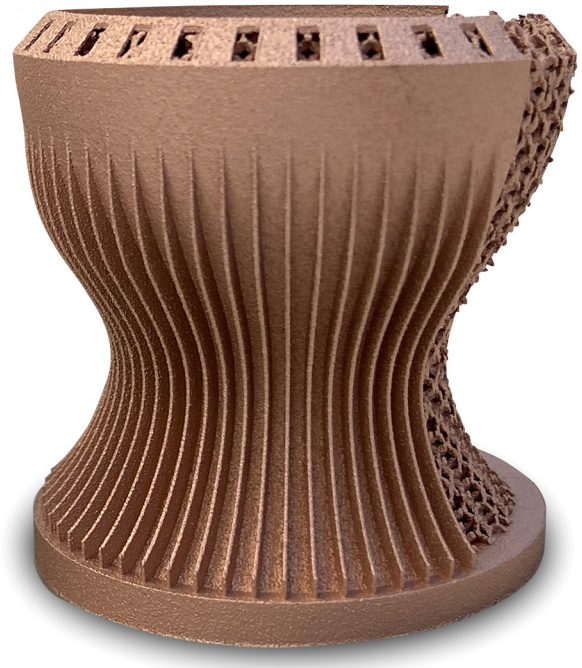
Material Data Sheet
Balanced Parameter



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Typical Mechanical Properties

Ultimate tensile strength	545 MPa
Yield strength	334 MPa
Elongation at break	25%

Process Parameter

Quality — Balanced — Productivity

Copper Based Alloy

A copper/chromium/niobium alloy developed by NASA specifically for use by spaceflight companies to manufacture components in rocket engines. The alloy exhibits high thermal conductivity combined with high strength at elevated temperatures and excellent creep resistance.

Highlights

- High thermal conductivity and strength at elevated temperatures
- Suited to parts in high temperature environments such as liquid rocket engines
- Applications in aerospace, space flight and satellites

- Layer thickness: 40µm
- Density >99.8%
- Good surface finish
- Dense and controlled thin walls <0.5mm

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Process Readiness Level (PRL)

R&D			Foundation			Production		
1	2	3	4	5	6	7	8	9

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Copper Based Alloy

To learn more, contact us: info@additiveindustries.com

or visit: additiveindustries.com

Powder Chemistry[2,3]

Composition	Cu	H	Si	Al	O	Fe	Nb	Cr	N
Min (wt%)	Bal.	Info Only	0	0	0	0	2.7	3.1	Info Only
Max (wt%)	-	-	0.035	0.06	0.07	0.025	3.0	3.4	-

Process details

Layer thickness	40	[µm]
Build rate[8] (per laser)	10.8	[cm3/hr]
Optical density [4]	≥ 99.8	[%]
Volumetric density [11]	-	[g/cm3]

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Specifications are subject to change without notice. Additive Industries is certified in accordance with ISO 9001

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Mechanical properties [5]	Orientation	As-built (Mean)	Standard Dev.	Heat-treat (Mean)	Standard Dev.	Units
Ultimate tensile strength	Horizontal	545	5	-	-	[MPa]
	Vertical	528	5	-	-	[MPa]
Yield strength	Horizontal	334	5	-	-	[MPa]
	Vertical	295	5	-	-	[MPa]
Elongation at break	Horizontal	25	2	-	-	[%]
	Vertical	27	1	-	-	[%]
Youngs modulus	Horizontal	127	15	-	-	[GPa]
	Vertical	142	10	-	-	[GPa]
Vickers Hardness [6]		143	5	-	-	[HV10]

Surface Roughness [7]	Mean	Standard Deviation	Units
Vertical Surface Roughness (Ra)	-	-	[µm]
Vertical Surface Roughness (Rz)	-	-	[µm]
45° Surface Roughness (Ra)	-	-	[µm]
45° Surface Roughness (Rz)	-	-	[µm]

Notes

1. The material is processed under Argon shielding atmosphere.
2. Powder Chemistry as per Additive Industries specification O8558 Rev. 1.0.
3. Additive Industries consolidated material is in compliance with industry specification (NASA)
4. Density measured by Optical Measurement Method as per internal process. This is the minimum guaranteed value that is achieved under standard processing conditions, manufactured using Additive Industries' qualification jobs.
5. Tensile test samples were produced as round blanks. These were machined to size and tested in accordance with ASTM E8m at a NADCAP approved supplier.
6. Hardness measured in accordance with DIN EN ISO6507-1:2018 as per internal process. Hardness values measured in XY and XZ planes from components manufactured using Additive Industries' qualification jobs.
7. Surface Roughness measured in as-printed condition in accordance with internal process. Roughness measurement conducted on specimens with varying unsupported manufacturing angle.
8. Build Rate stated is a typical value per laser. It is calculated using the formula: Layer Thickness x Laser Scan Speed x Hatch Distance.
9. Parameter released: GRCop42_40_BAL_MF1A64_INT_1.0.
10. Heat Treatment : NA
11. Volumetric density measured according to ASTM B962. This is a minimum volumetric density measurement achieved on samples manufactured using Additive Industries' qualification jobs.

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Disclaimer

The data presented in this material datasheet is valid only for Additive Industries' released powder, machine, and parameter sets, processed under the defined shielding atmosphere. The properties of the printed parts have been measured on test coupons according to industry standards where available, and the data correspond to our state-of-the-art at the time of publication. These results are based on Additive Industries' signoff build layout and reflect material performance under the specified conditions; for more information, please contact Additive Industries. Users should be aware that variations in the presented values may arise due to differences in process conditions, including but not limited to thermal management, build plate temperature, job-specific heat accumulation, inter-layer time, part positioning, and overall machine calibration. The data provided do not warrant any guarantee for printed parts, and it remains the responsibility of the producer or purchaser to verify the ultimate properties of the printed material for their specific application. The listed data are subject to change without notice as we continuously strive to develop and improve our machine performance and the properties of printed materials. Users are advised to exercise caution and consider material selection, build layout, and machine configuration when interpreting and applying this information.

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